A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-304			DATE:				
CUSTOMER: Air Asia	Company Ltd.	MCL 12,073 DTD					
CHANGE NO: 36							
TITLE: Specificat Coat Hook)	ion Administrative	Chan	ge (Revision to	Lavatory	r is		
ORIGIN: Convair in	itiated			т			
REASON FOR CHANGE:	To clarify the int	ent o	f the specificat	ion.			
EFFECT	ON WEIGHT *		EFFECT O	N BALANCE	*		
GUAR. WT. EMPTY	OPER, WT. EMPTY	# 1	0		INCH LB.		
EFFECT ON GUARANTEED I		143 0 23					
ETTECT ON GUARANTEED P	None			8 A			
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL	Ceally	7/7/K	PROVAL		
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:		CONVA	IR, A DIVISION OF GENER	AL DYNAMICS	CORP.		
BY:							
DATE:				EON V	AIR, S.D. 6-1217A		

Air Asia Company, Ltd. Change No. 36

Title: Specification Administrative Change (Revision to Lavatory

Coat Hook)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification.

Description of Change:

Page 84, Paragraph 3.19.2.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Revise below item as follows:

From: "One coat hook (folding-type on lavatory door)"

To: "One coat hook"

A DIVISION OF GENERAL DYNAMICS CORRORATION. SAN DIE GO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:				
CUSTOMER: Air Asia	Company, Ltd.		MCL 12,072 DTD				
CHANGE NO: 35			MODEL	22-4	(Convai	. "880")	
TITLE: Specifi 3.2.2.1	cation Administrat	tive C	hange	(Revisi	on to Par	ragraph	
ORIGIN: Convair	initiated						
REASON FOR CHANGE:	Specification cla	arific	ation.				
EFFECT	ON WEIGHT .			EFFEC	T ON BALAN	ICE .	
GUAR, WT. EMPTY	OPER, WT. EMPTY						
0	0				ò	INCH LB.	
EFFECT ON GUARANTEED P	ERFORMANCE: *						
	None	>	g				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT, UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:				AND	LUCY	47/, APPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFF	ECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:		CONVA	IR, A DIVIS	SION OF GEN	VERAL DYNAM	ICS CORP.	
BY:							
DATE:						PARTE II GAZIFA	

Air Asia Company, Ltd. Change No. 35

CONVAIR: SD

Title: Specification Administrative Change (Revision to Paragraph

3.2.2.1)

Origin: Convair initiated

Reason for Change: Specification clarification.

Description of Change:

Page 9, Paragraph 3.2.2.1 COLOR CODING:

Revise the paragraph as follows:

"COLOR CODING: All sections of plumbing shall be color coded and banded in accordance with Military Standards and, in addition, marking shall indicate the direction of flow and the purpose of the line, except for engine and pylon fluid system hoses. The part number of each tubing assembly shall be identified. Color bands and part number identifications shall be plastic tape in accordance with Standard AND 10375."

To: "COLOR CODING: All sections of plumbing, except engine and within fuel tanks, shall be color coded and banded. All plumbing shall indicate the direction of flow and the purpose of the line. The part number of each tubing assembly shall be identified. Color bands shall be plastic tape in accordance with Standard AND 10375."

A DITISION OF SENERAL DYNAMICS CORPORATION SAN DIESO, CALIFORNIA

3FEC NO. 20- 22-30-			DATE:		
CUSTOMER: Air Asia	Company, Ltd.		MCL 12,026	DTD 1	.6 May 1961
CHANGE NO: 34			MODEL: 22-4	(Conv	rair "880")
	ation Administrati No. 3 Pod)	ve Cha	nge (Change G	uantity	of Spare
ORIGIN Convair	initiated				
REASON FOR CHANGE:	Two of the spare connection with tem.				
EFFECT	ON WEIGHT '		EFFEC	T ON BAL	ANCE '
GUAR. WT. EMPTY	OPER. WT. EMPTY				
0	0			0	INCH LB.
EFFECT ON GUARANTEED	PERFORMANCE: None	е			
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND A FUTURE CHANGE PROPOSE HANGE IS DEPENDENT UPOTHE FOLLOWING CHANGES	SAL	Rec	ily 3	7.8/c/ [Josq 5]14/6
					IG APPROVAL
LATEST DATE OF ACCEPTA	ANCĖ:	AIRPLA	ANES AFFECTED.		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:		CONVA	IR, A DIVISION OF GER	NERAL DYN	AMICS CORP.
BY					
DATE:					"Olympia General Bury (Fig.

Air Asia Company, Ltd. Change No. 34

Title: Specification Administrative Change (Change Quantity of Spare

Wires to No. 3 Pod)

Origin: Convair initiated

Reason for Change: Two of the spare wires from No. 3 pod utilized in

connection with the main gear brake anti-skid sys-

tem.

Description of Change:

Page 53, Paragraph 3.16.5.4 WIRE ROUTING:

Change the last sentence in the paragraph to read as follows:

"Eight spare No. 20 wires shall be installed from the main circuit breaker to each pod, except No. 3 pod, which shall have six spare wires.

4 DIVISION OF GENERAL DYNAMICS CORPORATION - SAN DIEGO CALFORNIA

SPEC NO. 20- 22-304		DATE:					
CUSTOMER. Air Asia	Company, Ltd.	MCL 12,071 DTD					
CHANGE NO. 33	·	MODEL. 22-4 (Convai	r "880")				
TITLE: Specification Cab:	ation Administrati in Partition and M	e Change (Delete Literatu vable Coat Dividers)	re Pockets				
ORIGIN By AACL/	Convair mutual agr	ement.					
REASON FOR CHANGE:	As above						
EFFECT	ON WEIGHT *	EFFECT ON BALA	ANCE .				
GUAR, WT. EMPTY	OPER, WT. EMPTY						
0	0	0	INCH LB.				
EFFECT ON GUARANTEED F	PERFORMANCE: None		4.				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS	got the	G APPROVAL				
LATEST DATE OF ACCEPTA	NCE:	MRPLANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:		ONVAIR, A DIVISION OF GENERAL DYNA	AMICS CORP.				
BY.							
DATE			01 VAL 1, 4 (4, 12 4				

Air Asia Company, Ltd. Change No. 33

Title: Specification Administrative Change (Delete Literature Pockets

from Cabin Partition and Movable Coat Dividers)

Origin: By AACL/Convair mutual agreement.

Reason for Change: As above

Description of Change:

Page 81, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Revise the last sentence as follows:

"Literature pockets shall be installed on the aft face of both the main cabin forward partition, the movable coat compartments and directly below the food trays in the passenger seat backs."

To: "Literature pockets shall be installed in the passenger seat backs directly below the food trays."

A DIVISION OF SENERAL DYNAMICS CORPORATION SAIL OIL 40, CALIFORNIA

SPEC NO. 20- 22-304			DATE			
CUSTOMER: Air Asia Company Ltd.			MCL 12,070 DTD			
CHANGE NO. 32		MODEL: 22-4 (Convair "880")				
TITLE Specificati	on Administrative	Chang	ge (Landing Gear Pin Indicator			
ORIGIN: Convair ini	tiated					
REASON FOR CHANGE: TO	specify that sub	ject :	indicator lights are installed.			
EFFECT	ON WEIGHT .		EFFECT ON BALANCE *			
GUAR, WT. EMPTY	OPER. WT. EMPTY					
0	0		O INCH LB.			
EFFECT ON GUARANTEED P	PERFORMANCE: None					
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOS. ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL	GALLOLOS 4/1, WILL ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS.		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED BY.		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
DATE:			HONVELL, AND BUTTON			

Air Asia Company Ltd. Change No. 32

Title: Specification Administrative Change (Landing Gear Pin

Indicator Lights)

Origin: Convair initiated

Reason for Change: To specify that subject indicator lights are installed.

Description of Change:

Page 21, Paragraph 3.8.1.5 - LANDING GEAR SAFETY LOCK PINS:

Add the following to the end of the paragraph:

"Indicator warning lights shall be provided in the flight engineer's panel and shall be illuminated when the "safety lock" pins are inserted in the gear".

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:					
CUSTOMER: Air Asia Company, Ltd.			MCL 12,069 DTD					
CHANGE NO: 31		MODEL: 22-4 (Convair "880")						
TITLE: Specificat: provement (lon Administrative Changes)	Chang	ge (Miscellane	ous Desig	gn Im-			
ORIGIN: Convair in:	ltiated	- E		#1				
REASON FOR CHANGE: P	roduct improvement	<u> </u>						
EFFECT	ON WEIGHT *		EFFECT ON BALANCE *					
GUAR. WT. EMPTY	OPER. WT. EMPTY							
0	0	85	0		INCH LB.			
EFFECT ON GUARANTEED	PERFORMANCE: * None	11			a L			
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND FUTURE CHANGE PROPOSE HANGE IS DEPENDENT UPOF THE FOLLOWING CHANGES	SAL	gate	MINEERING	APPROVAL			
				TOINEERING	ALLKOVAL			
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	_ANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			. 0			
ACCEPTED:	* ***	CONV	AIR, A DIVISION OF GEI	NERAL DYNAM	NICS CORP.			
BY:					. 3			
DATE:					N VAIR, S.D. 6-1217A			

Air Asia Company, Ltd. Change No. 31

Page 1 of 2

Title:

Specification Administrative Change (Miscellaneous Design

Improvement Changes)

Origin:

Convair initiated

Reason for Change: Product improvement

Description of Change:

Page 3a, GENERAL ARRANGEMENT:

Replace above illustration in the specification with Enclosure (A) which shows Krueger Flap installation between inboard pylon and fuselage, and trailing edge inboard flap extension.

Page 26, Paragraph 3.10.1.3 RUDDER SYSTEM:

Insert the following sentence after the third sentence:

"In addition a hydraulically-operated servo unit powered from one hydraulic system shall be installed in the pedal linkage, in the nose wheel well, to reduce rudder pedal forces."

Page 27, Paragraph 3.10.2.1 LIFT AND DRAG INCREASING DEVICE SYSTEMS:

Delete the paragraph and substitute the following:

"LIFT AND DRAG INCREASING DEVICE SYSTEMS: The trailing edge flaps and leading edge slats and flaps shall be operated by hydraulic power controlled by a single lever on the pilot's pedestal. The leading edge slats and flaps shall be operated through torque tubes from a gear box which is powered by two hydraulic motors, one motor connected to each hydraulic system. The trailing edge flaps shall be operated through torque tubes from a gear box which is powered by either of two motors, one motor connected to each hydraulic system. By means of "no-back" type devices incorporated in the screw jacks, each control system shall be irreversible to prevent flap motion from airload. Full trailing edge flap deflection and leading edge slat and flap extension shall be obtainable from either hydraulic system. A monitoring system shall be provided to preclude asymmetrical positioning of the trailing edge flap in the event of mechanical failure. A pair of lights, one amber and one green, shall be installed on the center instrument panel for each right and left hand set of leading edge devices (slats and flaps). Lights shall illuminate "green" when the devices have fully extended and shall illuminate "amber" while the devices are in transit."

Air Asia Company, Ltd. Change No. 31

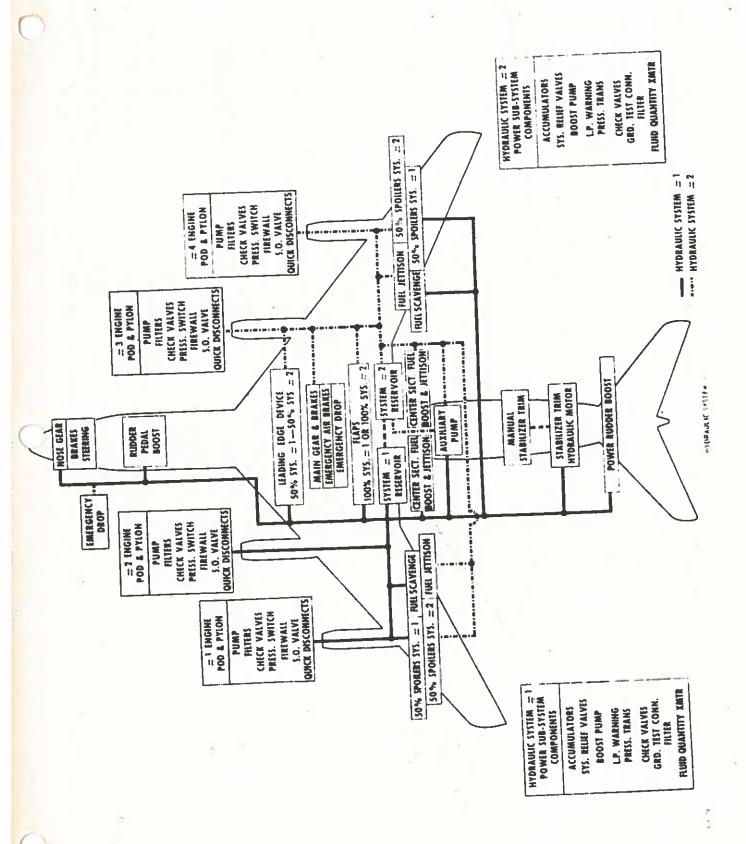
Page 2 of 2

Page 49a, HYDRAULIC SYSTEM:

Replace above illustration in the specification with Enclosure (B) which shows 100 percent flap power available from either No. 1 or No. 2 Systems, rudder pedal boost, center section fuel boost and jettison, and deletes "TEMP. CONTROL" callout.

Enclosure: (A) One copy Page 3a - GENERAL ARRANGEMENT (Revised)

Enclosure: (B) One copy Page 49a - HYDRAULIC SYSTEM (Revised)



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIL GO, CALIFORNIA

SPEC NO.: ZD- 22-304		DATE:			
CUSTOMER: Air Asia	Company, Ltd.	MCL 12,025 DTD 22 February 196			
CHANGE NO: 30		MODEL: 22-4 (Convair "880")			
TITLE: Buffets, W	eight Revision of				
		Nordskog Static Test Report" (undated)			
8	as defined by Nord	iskog Report.			
EFFECT (ON WEIGHT .	EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER. WT. EMPTY				
+94.0 lb	+94.0 lb	+6,500 INCH LB.			
EFFECT ON GUARANTEED P	ERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE COP Superseder No. 17.	FUTURE CHANGE PROPO ANGE IS DEPENDENT UPO HE FOLLOWING CHANGES	ON Attents 3/1/6, Wight			
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
3Y:					
DATE:		**************************************			
	W	UNITAL			

Title: Buffets, Weight Revision of

Origin: Convair initiated based on "Nordskog Static Test Report"

(undated)

Reason for Change: To show actual weight of buffets in specification

as defined by Nordskog Report.

Description of Change:

Page A-1, APPENDIX I-C, CUSTOMER FURNISHED - CONVAIR INSTALLED:

Revise the below item under Description as follows:

From: "4 Buffet 914.0"

To: "4 Buffet 1008.0"

Effect on Weight Empty: +94.0 pounds

Effect on Balance: +6,500 inch-pounds

Effect on Performance: None

The following shall not appear in the specification language:

The following are the buffet weights and capacities supplied by Nordskog:

Buffet No.		Weight (1b)	Capacity (1b)	Loaded Weight
1 2 3 4		389 211 219 189	355 504 525 419	744 715 744 608
	Total	1,008	1,803	2,811

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO.: 20- 22-304			DATE:			
CUSTOMER: Air Asia	Company, Ltd.	MCL 12,068 DTD				
CHANGE NO: 29			MODEL 22-4	(Convair	"880")	
	ation Administrati k Movement)	ve Ch	ange (Revisi	on to Pas	senger	
ORIGIN: Convair	initiated					
	To make specificat back positioning o			e with ac	tual seat	
EFFECT	ON WEIGHT .	er daug e	EFF	ECT ON BALA	NCE *	
GUAR, WT. EMPTY	OPER, WT. EMPTY					
0	0			0	INCH LB.	
EFFECT ON GUARANTEED F	ERFORMANCE: * None					
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	AL.	Ata	engineering	Ly luther GAPPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		RECUR NON-R	TON PRICE PER RRING: ECURRING:			
ACCEPTED:		CONVA	AIR, A DIVISION OF C	ENERAL DYNA	MICS CORP.	
BY:						
DATE:					or tall or or hands	

Air Asia Company, Ltd. Change No. 29

CONVAIR: SD

Title: Specification Administrative Change (Revision to Passenger

Seat Back Movement)

Origin: Convair initiated

Reason for Change: To make specification language agree with actual

seat back positioning capability.

Description of Change:

Page 81, Paragraph 3.19.1.1.6.1 SEAT BACK MOVEMENT:

Revise the paragraph to read as follows:

"Each reclining seat shall be designed so that pressure applied on the aft side of the seat back will override the recline control lock without additional manual operation, and fold the seat back forward to its normal upright position. This pressure on the seat back shall not exceed 25 pounds. The seat back shall fold to approximately 30 degrees forward of the vertical position by applying a load of not less than 30 pounds, or more than 35 pounds, at the top of aft side of the seat back without removing the center arm rest. The seat back may be further folded to approximately a horizontal position by removing the center arm rest. Seat back positions shall be as follows:

First Class Seats:

Normal Upright: 15 degrees aft of vertical

Maximum Recline: 38 degrees aft of vertical (With integral

folding food trays installed)

Recline: 38 and 45 degrees aft of vertical (With

integral folding food trays not

installed)

Coach Seats:

Normal Upright 15 degrees aft of vertical

Adjustable stop at: 23 and 38 degrees aft of vertical"

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304		DATE:				
CUSTOMER: Air Asia	Company Ltd.	MCL 12,067 DTD				
CHANGE NO: 28		MODEL: 22-4 (Convair "880")				
TITLE: Specification 3.10.1.5.1 a	n Administrative (nd 3.11.8)	hange ((Revision to Paragraphs			
ORIGIN: Convair init	iated					
REASON FOR CHANGE:	o clarify the inte	ent of a	subject paragraphs			
EFFECT	ON WEIGHT *		EFFECT ON BALANCE '			
GUAR, WT. EMPTY O	OPER. WT. EMPTY		O INCH LB.			
EFFECT ON GUARANTEED F	PERFORMANCE: *					
	None	4				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO		Attenty 161, WAS ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTA	NCE:	AIRPLAN	IES AFFECTED:			
SPECIAL PROVISIONS:		RECURRII	ON PRICE PER AIRPLANE: NG: URRING:			
ACCEPTED:		CONVAIR,	, A DIVISION OF GENERAL DYNAMICS CORP.			
DATE:			. "ONVAIN			

Air Asia Company Ltd. Change No. 28

CONVAIR: SD

Title: Specification Administrative Change (Revision to Para-

graphs 3.10.1.5.1 and 3.11.8)

Origin: Convair initiated

Reason for Change: To clarify the intent of subject paragraphs.

Description of Change:

Page 26, Paragraph 3.10.1.5.1 - STABILIZER TRIM CONTROL:

Delete the last two sentences and substitute the following:

"A switch, guarded in the on position, shall be installed on the pedestal to shut off all hydraulic power to the stabilizer screw jack. A three-position momentary switch shall be installed on the pedestal which, after the hydraulic shutoff switch is moved to "OFF"; operates the standby electrical-mechanical stabilizer trim control. In addition, hand cranks shall be installed on the pedestal for emergency manual stabilizer trim control".

Page 29, Paragraph 3.11.8 - INSPECTION AND MAINTENANCE:

Delete the last sentence and substitute the following:

"Each engine QEC (quick engine change build-up) shall be interchangeable and replaceable as a complete assembly, except for reverser cascades, and certain accessories which may not be required on individual installations".

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE 60, CALLEGRAIA

SPEC NO.: ZD-22-304			DATE:			
CUSTOMER: Air Asia	Company, Ltd.	MCL 12	MCL 12,066 DTD			
CHANGE NO: 27			MODEL: 22-4 (Convair "880")			
Specificat: 3.17.1.7.2	ion Administrative	Change (Re	vision to	Paragr	aph	
ORIGIN Convair in:	ltiated					
REASON FOR CHANGE: To	clarify the inte	it of Paragi	raph 3.17	7.1.7.2.		
	N WEIGHT .		EFFECT	ON BALAN	CE *	
GUAR, WT, EMPTY	OPER. WT. EMPTY					
0	0		0		INCH LB.	
EFFECT ON GUARANTEED P	ERFORMANCE: None			-		
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES: ENGINEERING APPROVAL					APPROVAL	
LATEST DATE OF ACCEPTAN	NCE:	AIRPLANES AFF	ECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED	ONVAIR, A DIVISI	ON OF GENE	RAL DYNAMI	ES CORP.		
BY:						
DATE				200	CATHOLOGI (61,17A	

Air Asia Company, Ltd. Change No. 27

CONVAIR: SD

Title: Specification Administrative Change (Revision to Paragraph

3.17.1.7.2)

Origin: Convair initiated

Reason for Change: To clarify the intent of Paragraph 3.17.1.7.2.

Description of Change:

Page 73, Paragraph 3.17.1.7.2 POWER CONTROLS:

Delete the paragraph and substitute the following:

"3.17.1.7.2 POWER CONTROLS: One "NORMAL" and one "ESSENTIAL" switch, located accessible to both pilots, shall control the ac and dc power to the radio equipment. The "NORMAL" switch shall control the weather radar, the marker beacon, selective call systems No. 1 and No. 2, glide slope No. 2, navigation instrument

transformer No. 2, VHF navigation No. 2, radio compass No. 2, VHF communication receiver No. 2, VHF communication transmitter No. 2, HF communication No. 2, DMET No. 1 and No. 2, if installed, ATC transponder No. 1 and No.2, if installed, and Doppler system No. 2, if installed. The "ESSENTIAL" switch shall control the VHF communication receiver No. 1, VHF communication transmitter No. 1, VHF navigation No. 1, glide slope No. 1, radio compass No. 1, navigation instrument transformer No. 1, HF communication No. 1, and the Doppler system No. 1. The emergency dc bus, through suitable circuit breakers, shall power the audio selector panels, the public address amplifier, the interphone, the pilot compartment speaker, the flight data recorder, and the tape

equipment shall be supplied by a separate ac and do line. All No. 2 normal radio equipment shall be supplied by a second separate ac and do line. Protection shall be provided for each line. The power supply, feeder wires, circuit protection for equipment which back up each other functionally shall be separated in a manner that will preclude operation

reproducer, if installed. All No. 1 normal radio

failure of both if any one circuit protective device feeder wire or power supply fails. Illumination shall be provided for the circuit breakers."

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIREGO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:					
CUSTOMER: Air Asia Company, Ltd.			MCL 12,065 DTD					
CHANGE NO: 26			MODEL: 22-4 (Conva	ir "880")				
TITLE: Specificati Landing Gea	on Administrative (r Tail Skid)	Chang	e (Revision to Auxi	liary				
ORIGIN: Convair ini	tiated							
REASON FOR CHANGE: P	roduct improvement							
EFFECT (ON WEIGHT *		EFFECT ON BA	LANCE *				
GUAR. WT. EMPTY	OPER. WT. EMPTY							
0	0		0	INCH LB.				
EFFECT ON GUARANTEED I	PERFORMANCE: * None			5				
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND FUTURE CHANGE PROPOSIANGE IS DEPENDENT UPON THE FOLLOWING CHANGES:	AL 1	Juleady	1/20/61				
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:					
EATEST BATE OF REGER IN								
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:						
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL D	YN AMICS CORP.				
BY:								
DATE:				7.0% V A114, S.D., 6-1217				

Air Asia Company, Ltd. Change No. 26

Title: Specification Administrative Change (Revision to Auxiliary

Landing Gear Tail Skid)

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page 23, Paragraph 3.8.3 AUXILIARY LANDING GEAR (TAIL SKID):

Change the first line in the paragraph to read as follows:

From: "A retractable skid or"

To: "A skid or"

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO. ZD- 22-304		DATE:					
CUSTOMER: Air As	ia Company Ltd.	MCL 12,064 DTD					
CHANGE NO. 25		MODEL: 22-4 (Convair "880")					
TITLE Specificat Transmitte	cion Administrati er)	ve Change (Relocation of Emergency					
ORIGIN: Convair in	nitiated						
REASON FOR CHANGE:	To make emergency all conditions in	transmitter readily available under compliance with FAA requirements.					
	ON WEIGHT *	EFFECT ON BALANCE *					
GUAR, WT. EMPTY	OPER. WT. EMPTY						
Negligible	Negligible	Negligible INCH LB.					
EFFECT ON GUARANTEED P	No						
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	FUTURE CHANGE PROPOS	SAL CARTINE 141					
LATEST DATE OF ACCEPTAN	ICE:	AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.					
BY:							
DATE:							
		CONVAIR,5.0. 8-1217A					

Air Asia Company Ltd. Change No. 25

Title: Specification Administrative Change (Relocation of

Emergency Transmitter)

Origin: Convair initiated.

Reason for Change: To make emergency transmitter readily available

under all conditions in compliance with FAA

requirements.

Description of Change:

Page 78, Paragraph 3.17.3.5.4 EMERGENCY RADIO TRANSMITTER:

In the third line, change "aft left hand stowage compartment." to "pilots' compartment."

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO. ZD- 22-304		D	DATE:			
CUSTOMER: Air Asia Company Ltd.		М	MCL 12,063 DTD			
CHANGE NO: 24			MODEL: 22-4 (Convair "880")			
	iministrative Cha	nge ((Revision to Engine Specifica-			
ORIGIN Convair initiate	ed					
REASON FOR CHANGE: To eng	specify the G.E. ines will be supp	Engir lied.	ne Specification under which			
EFFECT ON V	WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY OF	OPER. WT. EMPTY		o INCH LB.			
EFFECT ON GUARANTEED PER	FORMANCE: * None					
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		L	Heady 9 17/3/60 ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANC	Œ:	AIRPLA	ANES AFFECTED:			
RECL NON-			ECT ON PRICE PER AIRPLANE: URRING: -RECURRING: AL:			
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:			701 VAIR,5.0.0-12			

Air Asia Company Ltd. Change No. 24

COMMAIR: SD

1-1

Title: Specification Administrative Change (Revision to Engine

Specification)

Origin: Convair initiated

Reason for Change: To specify the G.E. Engine Specification under which

engines will be supplied.

Description of Change:

Page 2, Paragraph 2.2 - ENGINE SPECIFICATION:

Revise the paragraph as follows:

From: "General Electric Engine Specification E-723h dated

1 September 1959 shall apply".

To: "General Electric Engine Specification E-790 dated

11 July 1960 with Reverser Specification E-725e dated

11 July 1960 and Suppressor Specification E-731f dated

11 July 1960 shall apply".

Page 4, Paragraph 3.1.1.1 - PERFORMANCE:

In the ninth line, change "E-723h, dated 1 September 1959" to "E-790, dated 11 July 1960".

Page 101, Paragraph 3.23.2 - EQUIPMENT INTERCHANGEABILITY:

Under "a. Interchangeable Parts", add a double asterisk after "Engine (Quick Change) **" and add the following to the bottom of Fage 101:

"** NOTE: Except for reverser cascades".

Page A-3, APPENDIX I-C, PROPULSION EQUIPMENT:

Revise the following item under "POWER PLANT EQUIPMENT" as follows:

From: "4 Engine, complete *(Dry) Gen. Electric CJ-805-3B"

To: "4 Engine, complete *(Dry) Gen. Electric CJ-805-3B,

Spec. No. E-790 dated 7-11-60".

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DINGO, CALIFORNIA

CUSTOMER: Air Asia Company Ltd. CHANGE NO: 23		MCL 12,062 DTD MODEL: 22-4 (Convair "880")												
								TITLE Specification Administrative Change (Addition of Static Inverter to Paragraph 3.16.2.4.1)						
								ORIGIN. Convair initi	Lated					
	REASON FOR CHANGE:	Fo specify that st	atic	inverter is	s being fur	nished								
		ON WEIGHT *		EF	FECT ON BALAN	NCE 1								
	GUAR, WT, EMPTY	OPER. WT. EMPTY												
	0	0		INCH LB.										
EFFECT ON GUARANTEED PERFORMANCE: * None														
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			SAL .	d	Alasty ENGINEERING									
	LATEST DATE OF ACCEPTANCE:			AIRPLANES AFFECTED:										
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:												
ACCEPTED:			CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.											
	ВУ													
	DATE:					11. V 42. I. S 6-1 . 1 / A								

Air Asia Company Ltd. Change No. 23

CONVAIR: SD

Title: Specification Administrative Change (Addition of Static

Inverter to Paragraph 3.16.2.4.1)

Origin: Convair initiated

Reason for Change: To specify that static inverter is being furnished.

Description of Change:

Page 52, Paragraph 3.16.2.4.1 - POWER TRANSFER:

Add the following to the end of paragraph:

"A static inverter, connected to the battery, shall be provided for engine ground start when external a-c electrical power is not available"

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIF GO, CALIFORNIA

SPEC NO. ZD-22-304		DATE:						
CUSTOMER: Air Asia Company Ltd.		MCL 12,061 DTD						
CHANGE NO 22			MODEL: 22-4 (Convair "880")					
TITLE: Specification Administrative Change (Clarification of Cabin Ceiling Height)								
ORIGIN: Convair initiated								
REASON FOR CHANGE: To clarify cabin height in the area of the dropped ceiling.								
EFFECT	N WEIGHT *	22.39.23	EFFECT ON BALANCE *					
GUAR, WT. EMPTY	OPER. WT. EMPTY							
0	0		O INCH LB.					
EFFECT ON GUARANTEED P	ERFORMANCE: * None							
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES.			ENGINEERING APPROVAL					
LATEST DATE OF ACCEPTANCE:			LANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:						
ACCEPTED.		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.						
BY:								
DATE:			Approximate the second second					

Air Asia Company Ltd. Change No. 22

CONVAIR: SD

Title: Specification Administrative Change (Clarification of Cabin

Ceiling Height.

Origin: Convair initiated

Reason for Change: To clarify cabin height in the area of the dropped

ceiling.

Description of Change:

Page 86, Paragraph 3.19.3.1.1 - PASSENGER COMPARTMENT:

Revise the last sentence to read as follows:

"The clear ceiling height in the main cabin shall be approximately 85 inches at the airplane centerline except in the area of the dropped ceiling which shall be approximately 78 inches".



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304		DATE:	DATE:				
CUSTOMER: Air Asia	Company Ltd.	MCL 12,059 DTD	MCL 12,059 DTD				
CHANGE NO: 20		MODEL: 22-4 (Conv	mir "880")				
TITLE Specification Administrative Change (Revision to Paragraph 3.16.8.3.4 and to ADF Control Panel P/N)							
ORIGIN: Convair initiated							
REASON FOR CHANGE: To clarify the intent of the specification.							
EFFECT	ON WEIGHT *	EFFECT ON BA	ALANCE '				
GUAR, WT. EMPTY O	OPER. WT. EMPTY O	O	INCH LB.				
EFFECT ON GUARANTEED F	PERFORMANCE: * None						
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	Antead	19 178/60 RING APPROVAL				
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES AFFECTED:					
		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED COI		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.					
BY:							
DATE:			20% VA(1),, 0:1217A				

Title: Specification Administrative Change (Revision to Paragraph

3.16.8.3.4 and to ADF Control Panel P/N)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification

Description of Change:

Page 62, Paragraph 3.16.8.3.4 - READING LIGHTS:

Delete the last seven words in the fourth sentence, which reads as follows, and repunctuate sentence:

"and placarded to clearly indicate its function".

Page A-12, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "Control Panels" as follows:

From: "2 Radio Compass (ADF) Collins 614L-5"

To: "2 Radio Compass (ADF) Collins 614L-6"

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC No.: ZD- 22-304			DATE:				
CUSTOMER: Air Asia Company Ltd.			MCLDTD				
CHANGE NO: 19			MODEL: _	22-4	(Conv	air "88	0")
TITLE: Manufacturer	"s Name and/or Par	rt Num	ber.				
ORIGIN: Convair init	iated						
REASON FOR CHANGE:	o show current par quantities of Apper			nomen	elatur	e and	
EFFECT	ON WEIGHT *		Capper Access of the	EFFI	ECT ON	BALANCE	*
GUAR, WT. EMPTY	OPER. WT. EMPTY		0.				INCH LB.
EFFECT ON GUARANTEED	PERFORMANCE: * None						
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOS ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N		2	L	Cla	The
					ENGINE	ERING AP	PROVAL
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED: BY:		CONVA	AIR, A DIVIS	SION OF (GENERAL	DYN AMICS	
DATE:						CONVA	113,5.3. 6-1217A

Page

No

Air Asia Company Ltd. Change No. 19

1 of 7

Part or

Spec. No.

Total

Weight

Title: Manufacturer's Name and/or Part Number

Description

Origin: Convair initiated.

Reason for Change: To show current part numbers, nomenclature and

quantities of Appendix items.

Description of Change:

APPENDIX I-A, I-B and I-C EQUIPMENT:

Quan.

Reqd.

Change or Add manufacturer's names, part numbers or quantities to Items noted so as to read as follows:

Mfg. No.

	FURN	ISHING	38			
A-1	From:	4	Buffet			914.0
	To:	4	Buffet	Nordskog		914.0
	FUEL	SYST	EM EQUIPMENT			
A-3	From:	6	Pump, Jettison			
		2	Pump, Dual Purpose Supply- Jettison			
	<u>To</u> :	2 1 1 2 2	Pump, Jettison Pump, Jettison	Thompson Prod. Thompson	6191-6 6191-7	
		1	Jettison (Motor Assy) Pump, Dual Purpose Supply- Jettison (Motor Assy.)		206600-2	

Air Asia Company Ltd. Change No. 19

2 of 7

A-4	FLI	GHT INSTRUMENTS		Market Land
From:	*1	Indicator Longitudinal Trim Indicator, Flap Position (Dual)	U. S. Gauge Gen. Elec.	SR-012A 8DJ91LAB-Y1
<u>To</u> :	*1	Indicator Longitudinal Trim Indicator, Flap Position (Dual)	U. S. Gauge Gen. Elec.	SRL-012A 8DJ91LAB-Y1 or 8DJ91LAB-1
	SPE	RRY INTEGRATED FLIGHT SYSTEM		
From:	1	Vertical Gyro Flight Director Computer (Z-5)	Sperry Sperry	617926-1 615033 - 3
<u>To</u> :		Vertical Gyro Flight Director Computer (Z-5)	Sperry Sperry	258529 7-2 1780607
A-5	KOL	LSMAN INTEGRATED FLIGHT INSTRUMENT S	BYSTEM	
From:	#2	Altimeter and Scale Error Cor- rector Set	Kollsman	A32657-10-
	*4	Indicator, Exhaust Gas (Tempera- ture)	Gen. Elec.	8DJ100LAB
To:	*2	Altimeter and Scale Error Cor- rector Set	Kollsman	A32667-10-0
	#4	Indicator, Exhaust Gas (Tempera- ture	Gen. Elec.	8DJ100LAF
A-6	ENG	INE INSTRUMENTS - FUEL SYSTEM		
From:	2 2 2 1	Indicator, Fuel Quantity Totalizer Indicator, Fuel Quantity Counter Pointer Type Tanks 1 & 4 Indicator Refuel Quantity (1 & 4) Indicator Refuel Quantity (2 & 3) Indicator Fuel Quantity Counter Pointer Type Tanks 2 and 3 Indicator Fuel Quantity (CS) Indicator, Fuel Quantity Refuel (CS)	Simmonds	
To:	#1	Indicator, Fuel Quantity Totalizer	Simmonds	393032 - 02929
	5 5	Indicator, Fuel Quantity Counter Pointer Type Tanks 1 & 4 Indicator, Refuel Quantity (1 & 4) Indicator, Fuel Quantity Counter	Simmonds Simmonds	383053- 12919 383093-049
	2	Pointer Type Tanks 2 & 3 Indicator Refuel Quantity (2 & 3) Indicator Fuel Quantity (CS)	Simmonds Simmonds Simmonds	383053-129 383093-049 383053-139
	2	Indicator Fuel Quantity Refuel (CS)	Simmonds	383093-094

A-7

A-9

Air Asia Company Ltd. Change No. 19

3 of 7

AUTO PILOT TYPE SP-30

From:	1 Stabilization Computer 3 Amplifier, Servo 1 Computer, Flight Control 1 Yaw Command Computer 1 Relay Assembly	Sperry Sperry Sperry Sperry Sperry	1776002-3 619298-1 1776003-1 620134 1775387
To:	1 Stabilization Computer 2 Amplifier Servo Rudder Elevator 1 Amplifier, Servo-Aileron 1 Computer, Flight Control 1 Yaw Command Computer 1 Relay Assembly 1 Surface Servo Elevator 1 Surface Servo Aileron 1 Trim Coupler	Sperry Sperry Sperry Sperry Sperry Sperry Sperry Sperry Sperry	1776002-5 619298-4 619298-5 1776003-21 620134-3 1775387-3 615743-03 615743-10 1775390-7
From:	1 Roll Command Computer 1 Pitch Command Computer 1 Radio Coupler 1 Pressure Computer 1 Rack, Interleck 1 Indicator, Automatic Pilot 1 Gyro, Vertical 1 Trim Servo Sprocket and Bracket 3 Bracket, Servo 3 Servo Drive, Rudder	Sperry	1775339-2 1775391-2 1775388 1775882-4 1775392-1 1776004-1 617926-1 669352 615144-01 615743-03
<u> </u>	1 Roll Command Computer 1 Pitch Command Computer 1 Radio Coupler 1 Pressure Computer 1 Rack, Interlock #1 Indicator, Automatic Pilot 1 Gyro Vertical 1 Trim Servo Sprocket and Bracket 3 Bracket, Servo 1 Servo Drive, Rudder	Sperry	1775339-3 1775391-4 1775388-4 1775882-10 1775392-3 1776004-5 2585297-2 1780310 615144-01-1 615743-09
Add:	ELECTRICAL POWER EQUIPMENT 1 Static Inverter 1 Transformer Rectifier External Power	Gen. Elec. Electronic Specialty	352060DV101A

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A-10		ELECTRONIC EQUIPMENT		
		VHF NAVIGATION SYSTEM		
	From:	1 Antenna	Convair	
	To:	1 Antenna L/H 1 Antenna R/H	Convair Convair	30-35505-3 30-35505-4
		MARKER BEACON		
	From:	1 Receiver (1/4 ART)	Collins	51Z-2
	To:	1 Receiver (1/4 ART) (AC Power)	Collins	51Z-2
		VHF COMMUNICATION		
	From:	2 Transmitter 2 Receiver 1 Antenna (Fairing) 1 Antenna (Blade)	Collins Collins Convair Microwave Radiation	17L-7 51X-2
	To:	2 Transmitter (AC Power) 2 Receiver (AC Power) 1 Antenna (Fairing) 1 Antenna (Blade)	Collins Collins Convair Convair	17L-7 51X-2 22-30513 22-31006-3
		HF COMMUNICATION		
	From:	1 Antenna (Isolated Tail Cap)		
	To:	1 Antenna (Isolated Tail Cap)	Convair	22-14900-1
A-11		ATC TRANSPONDER BEACON	10	
	From:	1 Antenna	Electronic	LB-147
	To:	1 Antenna	Electronic Specialt	
		WEATHER RADAR	76	
	From:	1 Transmitter-Receiver	Bendix	RDR-1B-1
	To:	1 Transmitter- Receiver	Bendix	RDR-1B-2

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A-11		DO	PPLER NAVIGATION		
	From:	*1	Indicator	Bendix	INA-12A
	To:	*1	Indicator	Bendix	INA-12B-1
	Add:	1	Antenna Mount	Bendix	ADA-1202
A-12		CO	NTROL PANELS (ELECTRONIC)		
	From:	1243311	Weather Radar Radio Compass (ADF) Audio Selector Panel Microphone Selector Panel Public Address Panel ATC Transponder Beacon (Dual) HF Comm (Dual)	Gables Gables Gables Gables	614L-5 G-551 G-554
		1	Doppler Navigation	Gables	
	<u>To:</u>	1243111	Weather Radar Radio Compass (ADF) Audio Selector Panel Microphone Selector Panel Public Address Panel (Pilot) ATC Transponder Beacon (Dual) HF Comm (Dual)	Collins Gables Gables	G-565
		ī	Doppler Navigation	Gables	G-1023
	Add:	1	Public Address Panel (Fwd Stewardess) Public Address Panel (Aft Stewardess) Radio Power Radio Freq. Chart (HF)	Gables Gables Gables	G-816 G-815 G-635 G-587
A-14		PR	ESSURIZATION, ANTI-ICING AND AIR CO	NDITIONING	
	From:	2	Compressor, ATM Driven Cabin Evaporator, Freon	Ham. Stand Ham. Stand	the state of the s
	To:	2	Compressor, ATM Driven Cabin Evaporator, Freon	Ham. Stand Ham. Stand	
A-15		LA	NDING GEAR EQUIPMENT		
	From:	8822	Wheel, Main Brake, Main Wheel Wheel, Nose Brake, Nose Wheel	Bendix Bendix Bendix Bendix	
	To:	8882	Wheel, Main Brake, Main Wheel Wheel, Nose Brake, Nose Wheel	Bendix Bendix Bendix Bendix	152888 152889 152061 152062

Air Asia Company Ltd. Change No. 19

Controls

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A-16		FURNISHINGS		
<u>kr</u>	om:	3 Toilet, Flushing Type Dis- posal Tank Enclosures Wash		
		Stands and Hardware		
		4 Signs, "No Smoking-Fasten Seat Belt (Bilingual; English and		
		Chinese)		
		3 Signs, "Lavatory, Occupied-Vacant"		
		(Bilingual; English and Chinese) 3 Signs, "Return to Cabin" (Bilingual;		
		English and Chinese)		
To		Toilet Flushing Type (Aft Law L/W)	Monogram	2433
<u> </u>		Toilet, Flushing Type (Aft Lav L/H) Toilet, Flushing Type (Aft Lav R/H)		2434
	1	1 Toilet, Flushing Type (Fwd Lav		2483
		4 Signs, No Smoking-"Fasten Seat Belt"	Plastic Arts	
		(Bilingual; English and Chinese) Signs, Lavatory, "Occupied-Vacant"	Wilmond of Academ	00 00600
		(Bilingual; English and Chinese)	Plastic Arts	22-93009
	1-4-	3 Signs, "Return to Cabin"	Plastic Arts	22-93609
		(Bilingual; English and Chinese		
A - 17		AVUATRI BANT MINIM		
A-17	20 -	OXYGEN EQUIPMENT		
Fr	om:	3 *Cylinder, High Pressure Oxygen		
		(107 cu ft) (with gages) #Bottle, Portable Oxygen		
		(310 Liter)		
		Regulators (Crew Diluter Demand Type) Valve-Selector Oxygen Supplemental Oxygen Masks Smoke Masks	Water ware	16000
		3 Supplemental Oxygen Masks	Ewing Prod Scott	16000 Type A-1
	1100 3	3 Smoke Masks	Scott	-99- 11 -
То	:	3 *Cylinder, High Pressure Oxygen	ZEP	zc-268-1
		(107 Cu. Ft.) (With Gages)		
	3	3 *Bottle, Portable Oxygen (310 Litr) (Pass. Compt.)	Scott	55-00C1A BF20B
		3 Regulators (Crew Diluter	ARO	14950-8
		Demand Type) 3 Valve Selector Oxygen (Crew)	Dredmen Dread	76000
		3 Supplemental Oxygen Masks (Crew)	Ewing Prod. Scott	6849-2
		3 Smoke Masks (crew)	Scott	6849-9
Ad	đ:	l Valve-Oxygen Line	Robbins	0V601-1F
		1 *Bottle Portable Oxygen	Scott	6080-3
		(310 Liter) 1 Pressure Reducer (Crew)	Alar or	5758
			Consoli-	311W56
			dated Controls	
		1 *Pair Asbestos Gloves	Davis Condolidated	2214 FL
		2 Valve Auto Opening and Continuous	Controls	2020VJ

2 Valve Auto Opening and Continuous Flow Regulators (Pass)

SONVAIR: SD

Air Asia Company Ltd. Change No. 19

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A-17 (Cont) OXYGEN EQUIPMENT

Add: (Cont)

132 Oxygen Masks (Pass.)
10 Oxygen Masks (Pass.)
1 Restrictor-Fluid Flow Filter Scott 21236
6 Oxygen Mask Assy (Pass. Scott 5527 Compt).

Effect on Weight Empty: None Effect on Balance: None Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304		DATE:				
CUSTOMER: Air Asia	Company Ltd.		MCL 12,058 DTD			
CHANGE NO: 18			MODEL: 22-4 (Convair "880")	ł,		
THILE Specificat	ion Administrative	: Chan	nge (Revised C. G. Limits Grap	h)		
ORIGIN: Convair initiated						
REASON FOR CHANGE: T	o furnish an up to	date	e C. G. Limits Graph.			
EFFECT O	N WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER. WT. EMPTY		The second secon			
0	0		O INCH L	В.		
EFFECT ON GUARANTEED P		ne				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOR PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Charles Popular Popula			
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:						
DATE.						

CONVAIR: SD

Air Asia Company Ltd. Change No. 18

Title:

Specification Administrative Change (Revised C. G. Limits

Graph)

Origin:

Convair initiated

Reason for Change: To furnish an up to date C. G. Limits Graph.

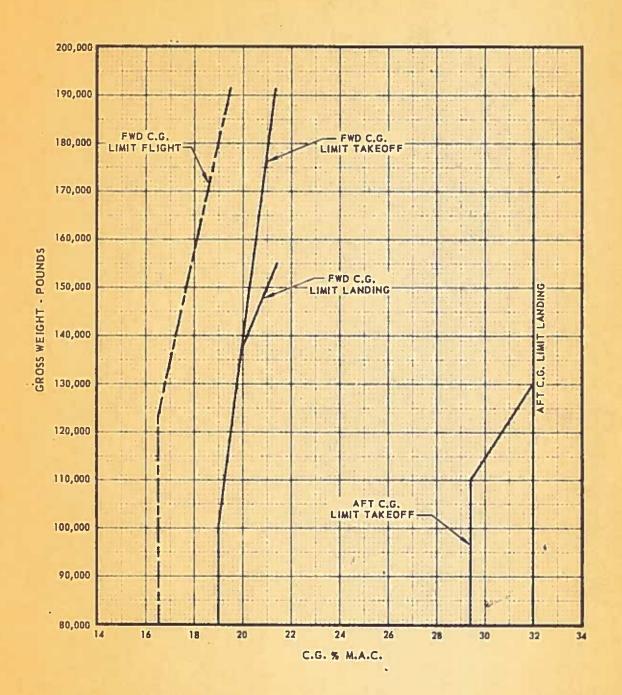
Description of Change:

Page 7a, C. G. DESIGN LIMITS:

Replace above illustration in the specification with Enclosure (A) which adds new aft C. G. Limit takeoff and landing.

Enclosure: (A) One copy Page 7a, C. G. LIMITS (Revised)

Effect on Weight Empty: O Effect on Balance: O Effect on Performance: None



C.G. LIMITS

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:	
CUSTOMER: _ Air Asia	Company Ltd.		MCL 12,021 DTD 29 August 1	96
CHANGE NO: 17			MODEL: 22-4 (Convair "880")	
TITLE: Buffet No	o. 3 Support Struc	ture,	, Additional Strengthening of	
ORIGIN: Verbal re 26 August	equest by Messers	Herol	ld and Gluskin of AACL on	
REASON FOR CHANGE:	Customer request			
	**			
EFFECT (ON WEIGHT *		EFFECT ON BALANCE *	
GUAR, WT. EMPTY	OPER. WT. EMPTY			
+31.0 lb	+31.0 lb		+38,564 INCH LB	
EFFECT ON GUARANTEED P	ERFORMANCE: *			
	Non	PARK ARREST		
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A	L BE ACCUMULATED AND FUTURE CHANGE PROPORTION	SAI	N. N.	
ACCEPTANCE OF THIS CHA			[1/1/den 1. 160	
PRIOR ACCEPTANCE OF TI	HE FOLLOWING CHANGES		allegerens 160	
			ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTAN	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS			CT ON PRICE PER AIRPLANE:	
		IAOIA-KE	RRING:ECURRING:	-
		TOTAL		
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.	
BY:				
DATE:				

CONVAIR: SD Air Asia Company, Ltd. Change No. 17

Title: Buffet No. 3 Support Structure, Additional Strengthening of

Origin: Verbal request by Messers Herold and Gluskin of AACL on

26 August 1960 '

Reason for Change: Customer request

Description of Change: .

Page A-1, APPENDIX I-C, CUSTOMER FURNISHED - CONVAIR INSTALLED:

Revise the below item under "Description" as follows:

From: "4 Buffet 914.0"

To: "4 Buffet 945.0"

Effect on Weight Empty: +31.0 pounds

Effect on Balance: +38,564 inch-pounds

Effect on Performance: None

The following shall not appear in the specification language:

This change proposes strengthening the support structure of Buffet No. 3 to accommodate a higher loaded weight of 743 pounds.

. CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: 20- 22-304		DATE:			
CUSTOMER: Air Asia	Company Ltd.	MCL 12,020 DTD 10 August 1960			
CHANGE NO: 16			MODEL: 22-4 (Convair "880")		
TITLE: .Electrica	l Connections to E	Buffet	ts, Changes to		
ORIGIN: Telecon, on 15 Jul		atives	s and Parks of Nordskog		
	o make electrical ustomer furnished		ections compatible with ets.		
EFFECT (ON WEIGHT *		EFFECT ON BALANCE *		
GUAR, WT. EMPTY	OPER. WT. EMPTY				
+2.0 lb +2.0 lb			+644 INCH LB.		
EFFECT ON GUARANTEED F	PERFORMANCE: * None				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	alogueur 60		
			ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:					
DATE:		*	CONVAIR, B.O. DATE:		

CONVAIR: SD

Air Asis Company Ltd. Change No. 16

Title: Electrical Connections to Buffets, Change to

Telecon, Convair representatives and Parks of Nordskog, Origin:

on 15 July 1960

Reason for Change: To make electrical connections compatible with

Customer furnished buffets.

Description of Change:

This proposal has no effect on specification language.

Effect on Weight Empty: +2.0 pounds Effect on Balance: +644 inch-pounds

Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE 50, CALIFORNIA

SPEC NO.: ZD- 22-304		DATE:						
CUSTOMER: Air Asia	Company, Ltd.		MCL 12,057 DTD_					
CHANGE NO: 15	A Annual Control		MODEL: 22-4 (Conve					
TITLE Specifica Well Clos	ation Administration sing Panels)	ve Ch	ange (Inclusion of I	Food Tray				
REASON FOR CHANGE: To	REASON FOR CHANGE: To indicate in specification that food tray well closing panels will be provided for last row of first class							
sea	seats and last row of coach seats.							
EFFECT	ON WEIGHT .	L. STATE	EFFECT ON BAI	LANCE *				
GUAR, WT. EMPTY	OPER. WT. EMPTY							
o	0			O INCH LB.				
EFFECT ON GUARANTEED P	EFFECT ON GUARANTEED PERFORMANCE: * None							
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			a avguenos					
			ENGINEER	RING APPROVAL				
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:					
SPECIAL PROVISIONS:		RECUP NON-R	CT ON PRICE PER AIRPLANE RRING: RECURRING: L:					
ACCEPTED:		CONV	AIR, A DIVISION OF GENERAL D	YNAMICS CORP.				
BY:	5.00							
DATE				94 NAT 124				

Air Asis Company, Ltd. Change No. 15

Title: Specification Administrative Change (Inclusion of Food Tray

Well Closing Panels)

Origin: Convair initiated

Reason for Change: To indicate in specification that food tray well

closing panels will be provided for last row of first class seats and last row of coach seats.

Description of Change:

Page 81, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Add the following after the eighth sentence which reads: "All passenger seats shall be equipped with receptacles for plug-in food trays":

"A food tray well closing panel shall be installed on the inboard seat back of last row first class seats and inboard seat back of last row coach seats."

Page A-16, APPENDIX I-C, FURNISHINGS:

Add the following items to the "Description List":

"2 Panels, Coach Food Tray Well Closing

2 Panels, First Class Food Tray Well Closing"

Effect on Weight Empty: O Effect on Balance: O Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE CO. CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:			
CUSTOMER: Air Asia C	ompany Ltd.		MCL 12,019 DTD			
CHANGE NO: 14			MODEL: 22-4 (Convair			
TITLE Main System	Hydraulic Pumps,	Chang	ge to			
ORIGIN: Convair in	tiated					
REASON FOR CHANGE:	Product improvement	t				
EFFECT (ON WEIGHT *		EFFECT ON BALA	NCE *		
GUAR. WT. EMPTY +31.0 1bs	OPER. WT. EMPTY +31.0 lbs		+23,091	INCH LB.		
EFFECT ON GUARANTEED F	PERFORMANCÉ: * None					
TOTALS REFLECTED IN A	FUTURE CHANGE PROPOS	SAL	Ollogueun 1/20/0	,0		
ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T						
			ENGINEERIN	G APPROVAL.		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		RECUR NON-R	TON PRICE PER AIRPLANE: RRING: ECURRING:			
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYN	AMICS CORP.		
BY:						
DATE:				1021 V A1 1,5, 6-1217 4		

CUNVAIR: SD

Air Asia Company Ltd. Change No. 14

Title: Main System Hydraulic Pumps, Change to

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page A-13, APPENDIX I-C, HYDRAULIC AND PNEUMATIC EQUIPMENT:

Revise the below item under "Description" as follows:

From: "4 Pump, Main System Ham. Standard"

(Variable Displacement

Type)

To: "4 Pump, Main System Vickers AS-61695-I

(Variable Displacement

Type)

Effect on Weight Empty: +31.0 lbs.

Effect on Balance: +23,091 in/lbs.

Effect on Performance: None

The following shall not appear in the Specification language:

The hydraulic pump will incorporate one electrical depressurization valve and one blocking valve. The bolt-on ports will be modified to suit the Model 22 installation, and the yoke blocked back to a 6 gpm rating in lieu of 10 gpm.

This proposal also deletes the four accumulators in the hydraulic return lines, since these accumulators are not required with the Vickers pump.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:		
CUSTOMER: Asiatic	Aeronautical Co., I	Ltd.	MCL 12,056		cember 1960 July 1960
CHANGE NO: 13A			MODEL: 22-4	("Convair	880")
TITLE: Specificatio Recorder, Re	n Administrative Ch	nange,	Waste King	Flight Da	ta
ORIGIN Convair init	iated				
c	o add detail part morder system. Avid 960 and Revision to	onics	letter to P		
EFFECT	ON WEIGHT *		EFF	ECT ON BALAN	ICE *
GUAR, WT. EMPTY	OPER, WT, EMPTY				
None	None		No	ne	INCH LB.
EFFECT ON GUARANTEED	PERFORMANCE: * None				
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND A FUTURE CHANGE PROPOS HANGE IS DEPENDENT UPOR THE FOLLOWING CHANGES:	SAL N	Albah	17176 ENGINEERING	
LATEST DATE OF ACCEPT	ANCE:	AIRPLA	NES AFFECTED	:	
SPECIAL PROVISIONS:		RECURF NON-RE	ON PRICE PER		
ACCEPTED:		CONVAI	R, A DIVISION OF	GENERAL DYNAM	AICS CORP.
BY.					
DATE:					NVA1165.D. 5-1217A

Asiatic Aeronautical Co. Ltd. Change No. 13A

Specification Administrative Change, Waste King Flight Data Title:

Recorder, Revision of

Origin: Convair initiated

Reason for Change: To add detail part numbers for the flight data records system. Avionics letter to P.O. dated 15 June 1960.

and Revision to CCP-13

Description of Change:

Page A-1, APPENDIX I-A - ELECTRONICS:

Change the below items as follows:

From:	"1	Flight Recorder	Waste King	5424 28.2
To:	1	Flight Data Recorder System Consisting of the following:	Waste King	5424- 200 32.0
		l Encoder (Trip and Date) l Magazine l Recording Unit	Waste King Waste King Waste King	10110 5427 10600-
		1 Acceleration Sensor	Waste King	5690

Effect on Weight Empty: None Effect on Balance: None Effect on Performance: None

A CIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:				(a. 15.0)
CUSTOMER: Asiatic A	eronautical (Co., Ltd.	MCL 1	2,016	DTD_	26 May 19	960
CHANGE NO: 10			MODEL:	22-4	(Conv	air "880")
	tion Administ Revision to		ange (Ma	ıximum Oy	perat	ional	
ORIGIN Convair i	nitiated						
REASON FOR CHANGE:	To de	efine the	actual d	design of	f the	airplane.	
FFFECT	ON WEIGHT *			EFFEC	T ON BA	LANCE *	
GUAR, WT. EMPTY	OPER, WT, EMPT	ТҮ					
0		0				0 INCH	LB.
EFFECT ON GUARANTEED	PERFORMANCE: *	None					
NEGLIGIBLE CHANGES WI TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF	A FUTURE CHANGE	E PROPOSAL ENT UPON	all	ogueno GI 2 VI		RING APPROVA	AL
LATEST DATE OF ACCEPT	A 1 Long Pol	Tares					-30W
LATEST DATE OF ACCEPT	ANCE:	AIRI	LANES AFF	FECTED:			
SPECIAL PROVISIONS:		RECI NON-	IRRING:	G:			
ACCEPTED:		CON	AIR, A DIVI	SION OF GEN	ERAL D	YNAMICS CORP.	
BY:							
DATE:						CONVAIN	5.1 17.4

Asiatic Aeronautical Co., Ltd. Change No. 10

Title: Specification Administrative Change (Maximum Operational

Altitude, Revision to)

Origin: Convair initiated

Reason for Change: To define the actual design of the airplane.

Description of Change:

Page 10, Paragraph 3.2.7 OPERATING ALTITUDE:

Change the paragraph to read as follows:

"The maximum operating altitude of the airplane shall be 41,000 feet, with an operating cabin pressure differential of 8.3 psi +.1 psi (See 3.20.3.1).

Page 98, Paragraph 3.20.3.1 DESIGN DIFFERENTIAL PRESSURE:

In the second line, change "8.2" to "8.3 psi +.1 psi".

In the seventh line, change "8.2" to "8.3 psi +.1 psi".

Page 99a, ESTIMATED CABIN PRESSURE SCHEDULE:

The above illustration will be revised to show effects of this change.

Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Change the below item as follows:

117 Indicator, Cabin Altitude From: and Differential Pressure

Kollsman B26395-10-004"

Indicator, Cabin Altitude To: and Differential Pressure

(P/N to be supplied)

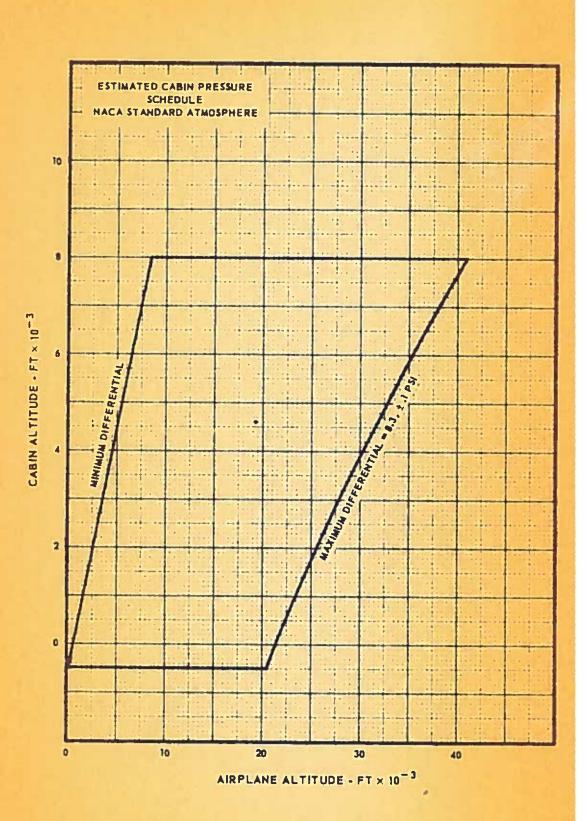
(A) One copy of Page 99a - ESTIMATED CABIN PRESSURE Enclosure: SCHEDULE (Revised to show effects of the above change)

Effect on Weight Empty: 0 Effect on Balance:

Effect on Performance:

None

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ESTIMATED CABIN PRESSURE SCHEDULE

3- N

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-304			DATE:		
CUSTOMER: Asiatic Ac	ronautical Co., Lt	td.	MCL 12,055	DTD 17	June 1960
CHANGE NO. 9			MODEL: 22-4	(Convair	"880")
	Administrative Ch	nange	(Specificati	on clarif	ication
ORIGIN Convair initi	ated	W1 5			J
REASON FOR CHANGE: TO	clarify intent of	the	Specification		
EFFECT O	N WEIGHT .		EFFE	CT ON BALAN	ICE *
GUAR. WT. EMPTY	OPER. WT. EMPTY				INCH LB.
Neg.	Neg.		Neg.		
EFFECT ON GUARANTEED PI	ERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	L BE ACCUMULATED AND FUTURE CHANGE PROPOS	SAL N	allogue 7/1	ur 2/60 Engineering	S APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:		D.
SPECIAL PROVISIONS:		RECUI NON-R	CT ON PRICE PER A RRING: ECURRING:		
ACCEPTED.		CONV	AIR, A DIVISION OF G	ENERAL DYNA	MICS CORP.
BY:					
DATE:					071 V 410, 5, 0, 6-1217

Asiatic Aeronautical Co., Ltd. Change No. 9

Page 2 of 4

Page 40, Paragraph 3.12.9.13.2 - EMERGENCY REFUELING:

Change the last sentence to read as follows:

"Means shall be provided adjacent to each filler for connecting a ground wire from the refueling nozzle".

Page 32, Paragraph 3.16.2.4.1 - POWER TRANSFER:

Change the last sentence to read as follows:

"To energize the ac busses from the ground power, the four bus tie switches shall be in the closed position, the ground power connected and the external power switch placed in the "ON" position".

Page 53, Paragraph 3.16.5.1 - GENERAL:

Change the third sentence to read as follows:

"Wire current ratings shall be guided by AIA document Design Manual for Aircraft Electrical Installations as revised May 1, 1952".

Page 56, Paragraph 3.16.7.1.2 - FUSES:

Change the paragraph starting with the third sentence to read as follows:

"Permanent labeling adjacent to each active fuse shall show the fuse rating in amperes and the device and/or circuit function which the fuse protects. Space provisions shall be provided on all fuse panel to permit installation of approximately ten percent additional fuses after finalization of panel original design. A supply of spare fuse conforming to CAR requirements shall be located so as to be accessible in flight. Spare fuse stowage shall indicate fuse rating. Spare fuses shall be provided in the following quantities: 100 percent spares for 12 or less active fuses of same rating per airplane. For more than 12 active fuses of same rating per airplane 50 percent spares or a minimum of 12 spare fuses (whichever is greater)".

Page 57, Paragraph 3.16.7.2 - PROTECTOR PANELS:

Change the last two sentences to read as follows:

"Space provisions including mounting holes for approximately ten percent additional circuit breaker shall be provided for the combined group of flight deck circuit breaker panels after finalization of panel original design".

CONVAIR: SD

Asiatic Aeronautical Co., Ltd Change No. 9

Page 3 of 4

Page 65, Paragraph 3.16.11.1 - LANDING GEAR AND STABILIZER WARNING HORN:

Change the last subparagraph to read as follows:

"When the airplane is on the ground and any one of the thrust levers is advanced to takeoff position, the horn will operate intermittently if the speed brakes are not retracted. When the airplane is on the ground and any two or more of the thrust levers are advanced to takeoff position, the horn will operate intermittently if the stabilizer and/or the flaps are not in the correct takeoff range".

Page 67, Paragraph 3.16.14 - BOOSTER COIL:

Delete the entire paragraph and revise heading to read

BOOSTER COIL: Not applicable.

Page 72, Paragraph 3.17.1.6 - RADIO JUNCTION PANELS:

Change last sentence to read as follows:

"All terminals in terminal strips shall be the steel nonrotating type, shall remain secured to the strips when terminal connections are removed and shall provide for removal of broken stude with a minimum of disruption of adjacent wiring".

Page 89, Paragraph 3.19.4.1.2 - SHUTOFF VALVES:

Change the paragraph to read as follows:

"Shutoff valves, electrically actuated by one handle located in the pilot compartment shall be provided for the fuel, hydraulic and blee air systems of each engine. (Reference 3.12.9.7.3, 3.15.1.6 and 3.16.16.)"

Page 102, Paragraph 3.23.2 - EQUIPMENT INTERCHANGEABILITY:

Change the following items in the Equipment List:

From: Crew and Passenger Seat Back Cushions and Arm Rest Covers.

To: Passenger Seat Assembly
Pilot and Copilot Seat Assembly
Flight Engineer Seat Assembly
Observer Seat Assembly

Page A-7 - APPENDIX I-C INSTRUMENTS AND RELATED EQUIPMENT:

Change the following items:

CONVAIR: SD

Asiatic Aeronautical Co., Ltd

Change No. 9

Page 4 of 4

1 Indicator Dual Cabin Ham. Standard 537325 From:

Supercharger Air Flow

1 Indicator, Dual Cabin To: 185249 Supercharger Air Flow Weston

Page A-8 APPENDIX I-C INSTRUMENTS AND RELATED EQUIPMENT:

Change the following item:

1 Indicator, Hydraulic Simmonds 393028-01645 From:

Fluid Quantity (Dual)

To: 1 Indicator, Hydraulic Fluid Quantity (Dual) Liquidometer EA148ANH-99H

Effect on Weight Empty: Neg. Effect on Balance: Neg. Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-304	DATE:
CUSTOMER: Asiatic Aeronautical Co., Lt	d. MCL 12,054 DTD 10 June 1960
CHANGE NO: 8	MODEL: 22-4 (Convair "880")
TITLE: Specification Administrati Lights Inoperative)	ve Change (Render Engine Overheat
ORIGIN: Convair initiated	
REASON FOR CHANGE: To prevent warning started or during t	light indication when engines are takeoff.
EFFECT ON WEIGHT *	EFFECT ON BALANCE *
GUAR, WT, EMPTY OPER, WT, EMPTY	
Neg. Neg.	Neg. INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: * None	e ·
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPORTION ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	ON allyguen 6/9/6
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	
DATE.	10° VA14,9,0, 6-(217

CONVAIR: SD

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Asiatic Aeronautical Co., Change No. 8

Title: Specification Administrative Change (Render Engine Overhead

Lights Inoperative)

Origin: Convair initiated

Reason for Change: To prevent warning light indication when engines

are started or during take-off.

Description of Change:

The following heas no effect on the specification language:

The engine overheat lights shall be rendered inoperative.

Page 43a, PILOT'S AND COPILOT'S INSTRUMENT PANEL:

Revise above illustration to reflect intent of this change.

Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO .: ZD- 22-304	The second section is a second		DATE:				
CUSTOMER: Asiatic A	eronautical Co. Lto	d.	MCL 12	053	_DTD_	9 Jun	e 1960
CHANGE NO: 7			MODEL:	22-4	("880")	
THILE Specification Addition of)		hange	(Speed	Brake	Warnin	g Hor	ก
ORIGIN: Convair Init	iated						
REASON FOR CHANGE: S	pecification corre	ction					
EFFECT ON WEIGHT 1				EFFEC	T ON BAL	ANCE '	i de la companya della companya della companya de la companya della companya dell
GUAR, WT. EMPTY	OPER. WT. EMPTY						
0	0			0			INCH LB.
EFFECT ON GUARANTEED	PERFORMANCE: None						
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND FUTURE CHANGE PROPOSIANGE IS DEPENDENT UPON THE FOLLOWING CHANGES:	SAL N	Wash		o NGINEERI	NG APP	ROVAL
LATEST DATE OF ACCEPTA	ANCE:	AIRPLA	NES AFFE	CTED:		N.	
SPECIAL PROVISIONS:		RECURI					
ACCEPTED:		CONVAI	IR, a DIVISI		NERAL DY		
БҮ:							
DATE						(2° × 4°)	12012 2 0 1 2 1 7 4

Aslatic Aeronautical Co. Ltd. Change No. 7

CONVAIR: SD

Title: Specification Administrative Change (Speed Brake Warning Horn,

Addition of)

Origin: Convair initiated

Reason for Change: Specification Correction.

Description of Change:

Page 66, Add the following new paragraph:

3.16.11.1.2 - SPEED BRAKE WARNING HORN: An intermittent warning horn shall be provided to sound whenever the spoiler are not retracted and any one thrust lever is advanced beyond the 92 percent rpm position.

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None * CONVAIR: SD

Asiatic Aeronautical Co., Ltd. Change No. 9

Page 1 of 4

Title: Specification Administrative Change (Specification Clarification

and Corrections)

Origin: Convair initiated

Reason for Change: To clarify intent of the specification.

Description of Change:

Page 11, Paragraph 3.4.2 - DESIGN SPEEDS (EQUIVALENT AIR SPEED):

Delete (EQUIVALENT AIR SPEED) from above paragraph title and add the following after the paragraph title:

"The airplane shall be designed to be operated at the following equivalent airspeeds:"

Add the following to the end of the paragraph:

"The airplane shall be placarded to be operated at the following indicated airspeeds:

Operating speed (flaps full down)	195 K or M = 0.40
Operating speed (flaps 3/4 down)	220 K or M = 0.50
Operating speed (flaps 1/2 down)	245 K or M = 0.60
Operating speed (main landing gear	
extension)	375 K or M = 0.89
Operating speed (nose landing gear	
extension	320 K or M = 0.83
Operating speed (landing gear re-	
traction)	270 K or M = 0.70
Operating speed (leading edge device	
extension	$245 \text{ K or M} = 0.60^{\circ}$

Page 38, add the following new paragraph:

"3.12.9.7.8 - DEFUEL VALVES: Two defuel valves shall be controlled from the pilot compartment. The valves shall be electrically actuated and located between the engine crossfeed line and the refuel manifolds. Provisions for electrical indication of opening shall be incorporated in each valve by means of an indicator light, which shall be "ON" when the valve is open".

A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIEGO, CALIFORNIA

3PEC NO.: 2D- 22-30	4		DATE:				
CUSTOMER: Asiatic A	eronautical Co., I	Ltd.	MCL 12.008 DTD Rev	. 29 March			
CHANGE NO: 6			MODEL: 22-4 (Convair "880")				
TITLE Interior Tr	im, Revision to	*					
ORIGIN: AACL Initiation 20, December	ted - Reference AA r 1959	CL Le	tter SA/P-59-160 dated				
REASON FOR CHANGE: CL	stomer Request						
	ON WEIGHT *		EFFECT ON BALAN	CE '			
+13.0 lbs.	OPER. WT. EMPTY		+6614	INCH L.B.			
EFFECT ON GUARANTEED P	ERFORMANCE: *						
	None						
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A	L BE ACCUMULATED AND FUTURE CHANGE PROPOS		of white				
ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	NGE IS DEPENDENT UPON HE FOLLOWING CHANGES:	1	(Workship				
	3*).		ENGINEERING A				
LATEST DATE OF ACCEPTAN	CE:	AIRPLA	ANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT	T ON PRICE PER AIRPLANE				
		RECUR	RING:				
			CURRING:				
ACCEPTED:		CONVAI	R, A DIVISION OF GENERAL DYNAMIC	The same of the sa			
3Y ₄							
DATE:							
				/al //4			

CONVAIR: SD

Asiatic Aeronautical Co., Ltd. Change No. 6

Page 1 of 2

itle: Interior Trim, Revision to

Origin: AACL Initiated - Reference AACL Letter SA/P59-160

dated 20 December

Reason for Change: Customer Request

Description of Change:

Page A-1, APPENDIX I-A - FURNISHINGS

Change the below items as follows:

From: AR Interior Furnishings Items (Medalion, Plaques, such as Dragon, etc.)

To: AR Interior Furnishings Items (Such as, Medalions, Plaques, Dragons, etc.)

Page A-19 APPENDIX I-D - FURNISHINGS UNIT WEIGHTS

Change the below items as follows:

Cabin Floor Covering (including under From: 53.0 oz/sq vd Carpet Pad Buffet Entryway and Lavatory Floor 32.0 oz/sq d Covering-Pwd (Total) 44.0 oz/sq 30 Buffet Entryway and Lavatory Floor Covering Aft Cabin Floor Covering (including under-carpet To: 57.0 oz/sq d Buffet Entryway and Lavatory Floor Covering-57.0 oz/sq yd Fwd (Total) Buffet Entryway and Lavatory Floor Covering-57.0 oz/sq yd Aft (Total)

Effect on Weight Empty: +13.0 lbs

Effect on Balance: +6,614 inch-1bs

Effect on Performance: None

CONVAIR: SD

Asiatic Aeronautical Co., Lt Change No. 6 Page 2 of 2

The following has no effect on the specification language and consists of installing the following items as described in the Customer's Interior Finish Specification

- 1. Install carpeting in forward and aft entryway of a different color than the main passenger compartment.
- 2. Install overhead grill work and bulkhead trim for coat divider partition and other partitions at station 403 and 1224.
- 3. Install laminated printed material for window blinds.
- 4. Anodize metal surfaces in lieu of bare aluminum finish on interior items previously agreed upon.

A DIVISION OF GENERAL DYNAMICS CORPONATION SAN DIE GO, CALIFORNIA

	SPEC NO.: ZD- 22-304	Contract of the Contract of th		DATE:					
	CUSTOMER Asiatic A	eronautical Co., L	td.	MCL 12.014 DTD	18 May 1960				
	CHANGE NO. 5	V.		MODEL: 22-4 (Conve	air "880")				
	TITLE: Fabric Rein.	forced Tires, Chan	ge to						
		AACL Initiated							
	REASON FOR CHANGE: P	roduct Improvement							
	GUAR WI EMPTY TOPER WI EMPTY			EFFECT ON B	ALANCE .				
	GUAR, WT. EMPTY	OPER. WT. EMPTY							
	+23.0 lbs.	+23.0 lbs.		+ 20,194	INCH LB.				
	EFFECT ON GUARANTEED P		one						
	* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPON	٧	alloqueun 6/14/60	RING APPROVAL				
	LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:					
	SPECIAL PROVISIONS:		RECUR NON-RI	T ON PRICE PER AIRPLANI RING: ECURRING: :					
7	ACCEPTED:		CONVA	IR, A DIVISION OF GENERAL D	YNAMICS CORP.				
	BY:								
	DATE:				Carles and and				

c CONVAIR: SD

Asiatic Aeronautical Co. Change No. 5

Title: Fabric Reinforced Tires, Change to

Origin: Convair and AACL initiated

Reason for Change: Product improvement

Description of Change:

Page A-14, APPENDIX I-C LANDING GEAR EQUIPMENT

Change The below items in the Equipment List as follows:

"8 Tire, Main Wheel (39 x 13) U. S. Rubber From:

(Type VII 22 Ply Rating-Skid Depth .25 inches)

"8 Tire, Main Wheel (39 x 13) To:

Type VII 22 Ply Rating, Fabric Reinforced Tread, Skid Depth

.35 inches) U. S. Rubber

Effect on Weight Empty: -23.0 lbs. Effect on Balance: -20,194 in. lbs. Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO.: ZD-22-304			DATE:				
CUSTOMER: Asiatic Ae	ronautical Co., Li	d.	MCL 12,015 DTD 18 May 1960				
CHANGE NO: 4			MODEL: 22-4 (Convair "880")				
TITLE: Specificati	on Administration	Chang	ge (Specification Correction)				
ORIGIN: Convair Ini	tiated						
REASON FOR CHANGE: M	emo R. R. Hoover t	o F.	S. Chambers dated 16 May 1960				
	ON WEIGHT *		EFFECT ON BALANCE *				
NONE	OPER. WT. EMPTY NONE		NONE INCH LB.				
EFFECT ON GUARANTEED P	PERFORMANCE: * NON						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			ENGINEERING APPROVAL				
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:		RECUR NON-RE	TON PRICE PER AIRPLANE: RING: ECURRING:				
ACCEPTED:		CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.				
BY:							
DATE:			0% / \$111, a, d, 44, 17A				

Asiatic Aeronautical Co., Lt CHANGE No. 4

TITLE:

Specification Administrative Change

(Specification Correction)

ORIGIN:

Convair Initiated

REASON FOR CHANGE: Memo R. R. Hoover to F. S. Chambers dated 16 May 1960

DESCRIPTION OF CHANGE:

Page 27, Paragraph 3.10.2.1 Lift and Drag Increasing Device Systems

DELETE: The seventh sentence in the paragraph which reads as follows:

> "Means shall be provided to warn the pilot that the flaps are extended and the airplane is operating at a speed that is in excess of the placard speed for that particular flap setting."

Effect on Weight Empty: Effect on Balance: Effect on Performance: None None None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

3FEC NO.: 2D-22-304			DATE:		-
CUSTOMER: Asiatic A	eronautical Co., Lt	d.	MCL 12,052	DTD_27	May 1960
CHANGE NO: 3			MODEL: 22-4	(Convair	"880")
TITLE Specificatio (Specificati	n Administrative Con Clarification)	Change			
ORIGIN: Convair Ini	tiateā	2000			
REASON FOR CHANGE: F	or Specification C	larif	ication and C	correction	1
	ON WEIGHT *		EFFE	CT ON BALAN	ICE *
NONE	OPER. WT. EMPTY NONE			NONE	INCH LB.
E FFECT ON GUARANTEED P	NONE				
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TI	FUTURE CHANGE PROPOS	AL	[[] [] [] [] [] [] [] [] [] [] [] [] []		
			E	NGINEERING	APPROVAL
LATEST DATE OF ACCEPTAN	NCE:	AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:		RECUR NON-RE	T ON PRICE PER AI RING: CURRING:		
ACCEPTED:		CONVAI	R, A DIVISION OF GER	NERAL DYNAM	ICS CORP.
BY:					
DATE:				100	(v t) (t, s.c., set,) (a

CONVAIR-SD

Asiatic Aeronautical Co., Lt. Change No. 2

TITLE: Specification Administrative Change (Specification Clarification)

ORIGIN: Convair Initiated

HEASON FOR CHANGE: For Specification Clarification and Correction

DESCRIPTION OF CHANGE:

Fage 31, Paragraph 3.19.1.1.6 Passenger Seats:

REVISE: The tenth sentence in the paragraph to read as follows:

FROM: "A rachet type footrest shall be attached to each passenger scat except for the last row of first class and coach seats."

To: "A rachet type footrest shall be attached to each passenger seat."

Page 92, Paragraph 3.19.7.2 Emergency Transmitter:

CHANGE: Change Paragraph Reference at end of sentence as follows:

FROM: " 3.17.3.5.3"

10: " 3.17.3.5.4"

Effect on Weight Empty: None Effect on Balance: None Effect on Performance: None Spor opular Cit

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO.: ZD- 22-304		DATE:				
CUSTOMER: Asiatic	eronautical Co. L	td.	MCL 12,05	1	DTD	
CHANGE NO: 1			MODEL: 2	2-4	(Convair	"880
TITLE Specification rectional arr	n Administrative C rows from Engine a	hange nd Pod	(Delete C	olor n Flu	Coding ar	nd Di-
ORIGIN: Convair Initi	ated					
REASON FOR CHANGE: Ge	eneral Electric sta ecessary	ates t	hat these	feat	ures are	un
EFFECT	ON WEIGHT *		Е	FFECT	ON BALANC	E *
GUAR, WT, EMPTY O	OPER. WT. EMPTY O			0		INCH
EFFECT ON GUARANTEED P						
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL	M	PO P	1/28/100 INEERING AI	PPROV
LATEST DATE OF ACCEPTA	NCE:	AIRPLA	ANES AFFECTE	ED:		
SPECIAL PROVISIONS:		RECURI NON-RE	T ON PRICE PERING:			
ACCEPTED:		CONVAI	R, A DIVISION O	F GENE	RAL DYNAMIC	s CORP.
BY:						
DATE:					* eps. v	Ki Barri